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Simulation study for the Stratospheric Inferred Winds (SIW) sub-millimeter limb sounder

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Abstract. Stratospheric Inferred Winds (SIW) is a Swedish mini sub-millimeter limb sounder selected for the 2nd InnoSat platform, with launch planned for around 2022. It is intended to fill the altitude gap between 30 and 70 km in atmospheric wind measurements and also aims at pursuing the limb observations of temperature and key atmospheric constituents between 10 and 90 km when current satellite missions will probably come to an end. Line-of-sight winds are retrieved from the Doppler shift of molecular emission lines introduced by the wind field. Observations will be performed with two antennas pointing toward the limb in perpendicular directions in order to reconstruct the 2-D horizontal wind vector. Each antenna has a vertical field of view (FOV) of 5 km. The chosen spectral band, near 655 GHz, contains a dense group of strong O₃ lines suitable for exploiting the small amount of wind information in stratospheric spectra. Using both sidebands of the heterodyne receiver, a large number of chemical species will be measured, including O₃ isotopologues, H₂O, HDO, HCl, ClO, N₂O, HNO₃, NO, NO₂, HCN, CH₃CN and HO₂. This paper presents a simulation study that assesses measurement performance. The line-of-sight winds are retrieved between 30 and 90 km with the best sensitivity between 35 and 70 km, where the precision (1 σ) is 5–10 ms⁻¹ for a single scan. Similar performance can be obtained during day and night conditions except in the lower mesosphere, where the photo-dissociation of O₃ in daytime reduces the sensitivity by 50 % near 70 km. Profiles of O₃, H₂O and temperature are retrieved with high precision up to 50 km (< 1 %, < 2 %, 1 K, respectively). Sys-

tematic errors due to uncertainties in spectroscopic parameters, in the radiometer sideband ratio and in the radiance calibration process are investigated. A large wind retrieval bias of 10–30 ms⁻¹ between 30 and 40 km could be induced by the air-broadening parameter uncertainties of O₃ lines. This highlights the need for good knowledge of these parameters and for studying methods to mitigate the retrieval bias.

1 Introduction

Millimeter and sub-millimeter (MM and SMM) limb sounders have been successfully used for more than 2 decades to probe the atmospheric composition and the temperature from the upper troposphere to the lower thermosphere (Waters et al., 1993; Murtagh et al., 2002; Waters et al., 2006; Kikuchi et al., 2010). The first generation of Millimeter Limb Sounder (MLS) provided unique observations of ClO, O₃, H₂O and HNO₃ allowing, for instance, a better understanding of the physical and chemical processes leading to the northern high-latitude O₃ depletion (Waters et al., 1993). Subsequent SMM limb sounders have enabled monitoring of the middle atmosphere (15–110 km) almost without interruption since the first MLS and have significantly contributed to the current middle-atmospheric measurement database (Hegglin and Tegtmeier, 2017). However, no successor missions are currently planned, and there is a risk of an observation gap in the near future (Livesey and Santee, 2017).

The benefits of such observations are manifold. The thermal emission spectrum at MM and SMM wavelengths is rich in isolated spectral lines from asymmetric molecules and molecular oxygen. Some important chemical species, such as HO₂ and ClO, have their clearest signals in this region of the spectrum (Urban et al., 2005; Khosravi et al., 2013; Sagawa et al., 2013; Millán et al., 2015). The O₂ lines give temperature and pressure, and the limb geometry provides a suitable vertical resolution for describing the middle atmosphere. Molecules are sensed in the thermal equilibrium state with no diurnal difference in the measurement performance, and measurements are not perturbed by stratospheric polar clouds and aerosols. Furthermore, the technology is mature, allowing sounders to operate over a period of longer than a decade. Methods have already been developed for improving the horizontal resolution with tomographic observations (Livesey et al., 2006; Christensen et al., 2015) and for obtaining very high signal-to-noise ratios using 4 K cryogenic cooling (Kikuchi et al., 2010).

Modeling middle-atmospheric major dynamical phenomena such as vertically propagating tidal waves, high-latitude sudden-stratospheric warming and equatorial quasi-biennial oscillation is still challenging (Limpasuvan et al., 2012; Newman et al., 2016; Orsolini et al., 2017; Sakazaki et al., 2018). Wind is one of the primary parameters for describing the physical state of the atmosphere, but models have difficulties in reproducing it where the atmospheric flow cannot be described by the geostrophic approximation, such as in the equatorial region, where the Coriolis force is weak and in the upper stratosphere and mesosphere, where waves and tides tend to dominate the wind fields (Baron et al., 2013b; Le Pichon et al., 2015; Kawatani et al., 2016; Duruisseau et al., 2017; Rüfenacht et al., 2018). As climate and weather models increase their vertical range to encompass more of the stratosphere and mesosphere, the need for measurements to improve the accuracy of models in this region, and hence at lower altitudes, can be expected to rise (Baldwin et al., 2003, 2010; Hoppel et al., 2008; Gerber et al., 2012).

Only the High Resolution Doppler Imager (HRDI) on the Upper Atmosphere Research Satellite (1991–2005) was able to measure horizontal winds over the stratosphere and mesosphere (Ortland et al., 1996), and current spaceborne sensors are not able to measure wind accurately below 90 km (Shepherd, 2015).

Ground-based stations do not cover the globe uniformly and most of the data are limited to heights below 30 km (Ishii et al., 2017) or above 70 km (Baumgarten, 2010). However recent efforts have been made to close this altitude gap (Baumgarten, 2010; Rüfenacht et al., 2014; Le Pichon et al., 2015; Blanc et al., 2018).

Providing wind data in the middle atmosphere from space is one of the challenges for future missions. In August 2018, the European Space Agency will launch the Atmospheric Dynamics Mission (ADM) Aeolus equipped with a wind lidar to demonstrate the feasibility of such measure-

ments (Stoffelen et al., 2005). However lidar is well suited for measuring wind in the troposphere but has poor precision above 20–30 km (Ishii et al., 2017). The Stratospheric Wind Interferometer For Transport studies (SWIFT) instrument has been proposed by the Canadian Space Agency for deriving winds between 15 and 45 km from O₃ infra-red emission lines (Rahnama et al., 2013). The launch was originally planned for 2010 but it is now very uncertain.

The potential of MM/SMM limb sounders for measuring winds has been demonstrated with line-of-sight wind retrievals between 70 and 90 km from the MLS O₂ line (Wu et al., 2008) and between 30 and 80 km from O₃ and HCl lines measured with Superconducting Submillimeter-Wave Limb-Emission Sounder (SMILES) (Baron et al., 2013b). Wind is one of the main parameters studied by SMILES-2 that has been proposed to the Japan Aerospace Exploration Agency (JAXA) (Ochiai et al., 2017). SMILES-2 is a large instrument (> 500 kg), which uses cryogenic SMM and THz receivers designed for very sensitive observations between 15 and 150 km. If the project is approved this year, the launch will be near 2025. Two smaller instruments are under consideration for satellite launch as early as 2020–2022. Wu et al. (2016) propose a small instrument for measuring the atomic oxygen line at 2.06 THz in order to retrieve its abundance as well as temperature and wind in the lower thermosphere. However this instrument cannot provide stratospheric and lower mesospheric information. The second proposal is the Stratospheric Inferred Winds (SIW) instrument. It is a small and low-cost satellite instrument being studied as part the Swedish InnoSat program (Lindberg, 2016; Murtagh, 2016). This program has planned to launch a scientific instrument every 2 years, and SIW has been selected for the second launch near 2022. It will observe the middle atmosphere (15–90 km) for a period expected to be at least 2 years, and will provide horizontal wind vectors within 30–90 km. The other primary products are the profiles of temperature, O₃, H₂O and more than a dozen of other chemical species. With this mission it will be possible to ensure the continuous monitoring of the middle atmosphere, avoiding an SMM measurement gap.

In this paper we present a simulation study that assesses the potential of SIW. A special focus is put on the main parameters: wind, temperature, O₃ and H₂O that are derived from the strongest lines in the selected spectral bands. Section 2 describes the mission and the observation technique. The measurement simulation and the retrieval method are explained in Sects. 3 and 4, respectively. The measurement performance is discussed in Sect. 5 and concluding remarks are given in the final section.

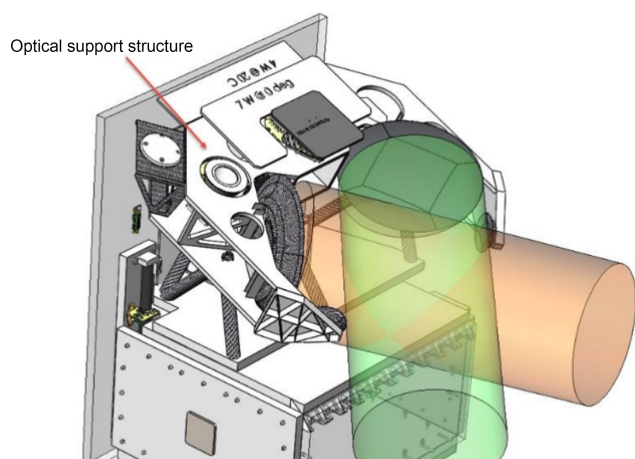


Figure 1. View of the InnoSat satellite. The box in the lower part is the platform service module. Above it is the scientific payload including the two antennas. The field of views are represented with green and beige colors (from Omnisys Instruments).

2 Mission description

2.1 Observation and instrument characteristics

The scientific payload (Fig. 1) and observation characteristics are summarized in Table 1. This is the proposed setting which can still be slightly modified. The platform will be set at a sun-synchronous polar orbit at an altitude of 550 km. It will fly near the terminator, crossing the equatorial ascending node at 18:00 local time (LT). Atmospheric observations will be performed toward the night side using two antennas facing perpendicularly to each other with angles from the satellite velocity of 45 and 135°, respectively. The antennas will point toward close air masses with few minutes delay (Fig. 2). They are fixed on the platform and the whole satellite will nod up and down in order to scan the limb alternatively upward and downward from about 15 to 90 km. The forward antenna is used during the upward scans and the aftward one during the downward scans. With this choice, the horizontal displacement of the tangent point during a vertical scan is less than 300 km, the vertical motion of the line of sight partly counterbalancing the satellite motion. Using the line-of-sight (LOS) winds retrieved with the two antennas over close regions allows us to derive the meridional and zonal wind components (Appendix A). The separation between the LOS wind profiles is less than 400 km.

The signals from the antennas are alternatively sent to a single radiometer passively cooled to 70 K below the ambient temperature, and analyzed with an autocorrelator spectrometer. The heterodyne radiometer operates in double-sideband (DSB) mode yielding to the superposition in the measured spectrum of the two image bands with respect to the local oscillator (LO). The bandwidth and resolution are 8 GHz and 1 MHz, respectively.

Table 1. Characteristics of the SIW payload and observations. The relationship between tangent height and LOS angle is derived for an Earth radius of 6370 km and a satellite altitude of 550 km above the geoid.

Payload dimensions	40 × 70 × 40 cm
Payload mass/power	17 kg/47 W
Antenna diameter	30 cm
Satellite altitude	500–600 km
Orbit inclination	98° (sun synchronous)
Latitude range	65° S–82° N
Local time of ascending node	18:00
Scan altitude	10–90 km
LOS nadir angle	67.25–69.03° (1.78°)
Scan velocity	0.05 ° s ^{−1} (35 s scan ^{−1})
Spectrum integration time	0.5 s (1.14 km)
Antenna vertical FOV	5 km
DSB system temperature	1000–1200 K
ACS Bandwidth	8 GHz
ACS resolution	1 MHz
LO frequency	638.075 GHz ($\lambda = 0.47$ mm)
IF frequency	10.975–18.975 GHz
Frequency \leftrightarrow velocity	1 m s ^{−1} \leftrightarrow 2 kHz

* Tangent point vertical displacement.

The strategy for acquiring the calibration data has yet been definitively decided upon, and it will probably be optimized in the future. Currently the plan is as follows. A calibration load onboard the platform (black body at ambient temperature) is viewed at the bottom and top of each scan during the turnaround. While limb scanning, the atmosphere and cold sky are observed alternatively with an integration time of 0.5 s each. Hence, atmospheric spectra are obtained every 2.3 km with an effective vertical resolution of about 5 km.

2.2 Spectral bands

The measured spectrum is composed of molecular lines spectrally resolved (Fig. 3). Using a radiative transfer model, they are inverted to retrieve geophysical information. Volume mixing ratio (VMR) and temperature are retrieved from their amplitude, whereas pressure and line-of-sight wind are retrieved from the width and the frequency position of the lines, respectively.

The Doppler shift induced by the LOS wind (2 kHz for 1 m s^{−1}) is small compared to the line broadening (1–100 MHz). This gives a very weak signal to exploit, especially in the lower stratosphere. Baron et al. (2013a) have shown that the spectral region near 655 GHz is the most suitable for measuring wind with the current hardware. It contains a dense group of strong O₃ lines (second row of Fig. 3), that increases by at least a factor of 2 the wind measurement sensitivity between 40 and 70 km compared to retrievals performed with a band of similar characteristics but located at

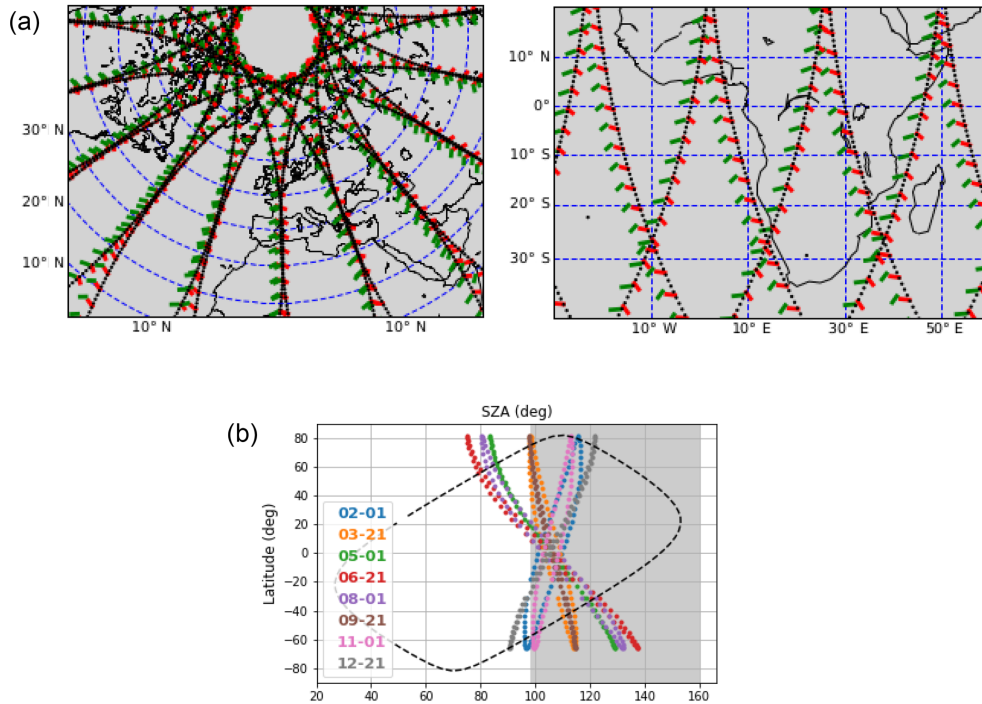


Figure 2. Panel (a) shows the footprints of the forward (45°) and aftward (135°) views over a 24 h period. The forward antenna is used during the upward scans (red lines) and the aftward one during the downward scans (green lines). The first tangent point of the upward scans are located on the black-dotted lines. Panel (b) shows the solar zenith angles with respect to latitudes for various days representative of the seasonal variation (colored dots) together with those of the Aura MLS data (DJF, 2011) used in the simulations (dashed line). The shaded area shows the nighttime measurements in the mesosphere, where the O_3 diurnal variation is the strongest.

any other frequency under 800 GHz. This band also allows us to retrieve temperature with good precision in the stratosphere without measuring an O_2 line.

The local oscillator frequency has been carefully selected in order to include as many as possible spectral lines and to reduce the line superposition from both sidebands. Hence lines of chemical species such as HCN (620.3 GHz), $H^{37}Cl$ (625.0 GHz), $H^{35}Cl$ (625.9 GHz), ^{35}ClO (649.5 GHz), NO (651.1 GHz) and N_2O (652.8 GHz) are clearly visible. A strong H_2O line is located at 620.7 GHz but very close to an O_3 line with similar strength. Lines from around twenty molecules are available though some are very weak such as H_2CO , CH_3Cl or BrO . Finally let us note that most of the lines where intermediate frequency (IF) of > 14 GHz have already been measured with Aura MLS and JEM/SMILES.

3 Measurement modeling

3.1 Radiative transfer and instrument

The signal is a spectral and spatial average of specific intensities ($W m^{-2} sr^{-1} Hz^{-1}$) over narrow instrumental functions. It is expressed in the so-called Rayleigh–Jeans bright-

ness temperature T_b equal to (Urban et al., 2004)

$$T_b(\theta_j, \vartheta_k) = \kappa_b \int_{\Delta\vartheta} d\vartheta g^{sp}(\vartheta - \vartheta_k) \left\{ w_{lsb}(\vartheta) \int_{\Delta\theta} G_e^{ant}(\theta - \theta_j, \nu_{lo} - \vartheta) I(\theta, \nu_{lo} - \vartheta) d\theta \right. \\ \left. + (1 - w_{lsb}(\vartheta)) \int_{\Delta\theta} G_e^{ant}(\theta - \theta_j, \nu_{lo} + \vartheta) I(\theta, \nu_{lo} + \vartheta) d\theta \right\}, \quad (1)$$

where ϑ_k is the frequency of the k th spectral component of the measurement, θ_j is the mean nadir angle during the measurement integration time of the j th spectra of the scan, and I is the specific intensity. The specific intensity is integrated along a LOS as that shown in Fig. 4. The LOS is characterized by the altitude of the tangent point, the angle with the meridional direction ϕ_n and narrow ranges over which the atmosphere is considered homogeneous. The heterodyne receiver is sensitive to atmospheric radiation at frequencies $\nu_{lo} \pm \vartheta$, where ν_{lo} and ϑ are the local oscillator and intermediate frequencies (Table 1). The instrumental functions are the spectrometer channel response g_{sp} (Hz^{-1}), the relative weight of the radiometer sidebands w_{lsb} and the effective antenna pattern G_e^{ant} . The parameter κ_b is the Rayleigh–Jeans

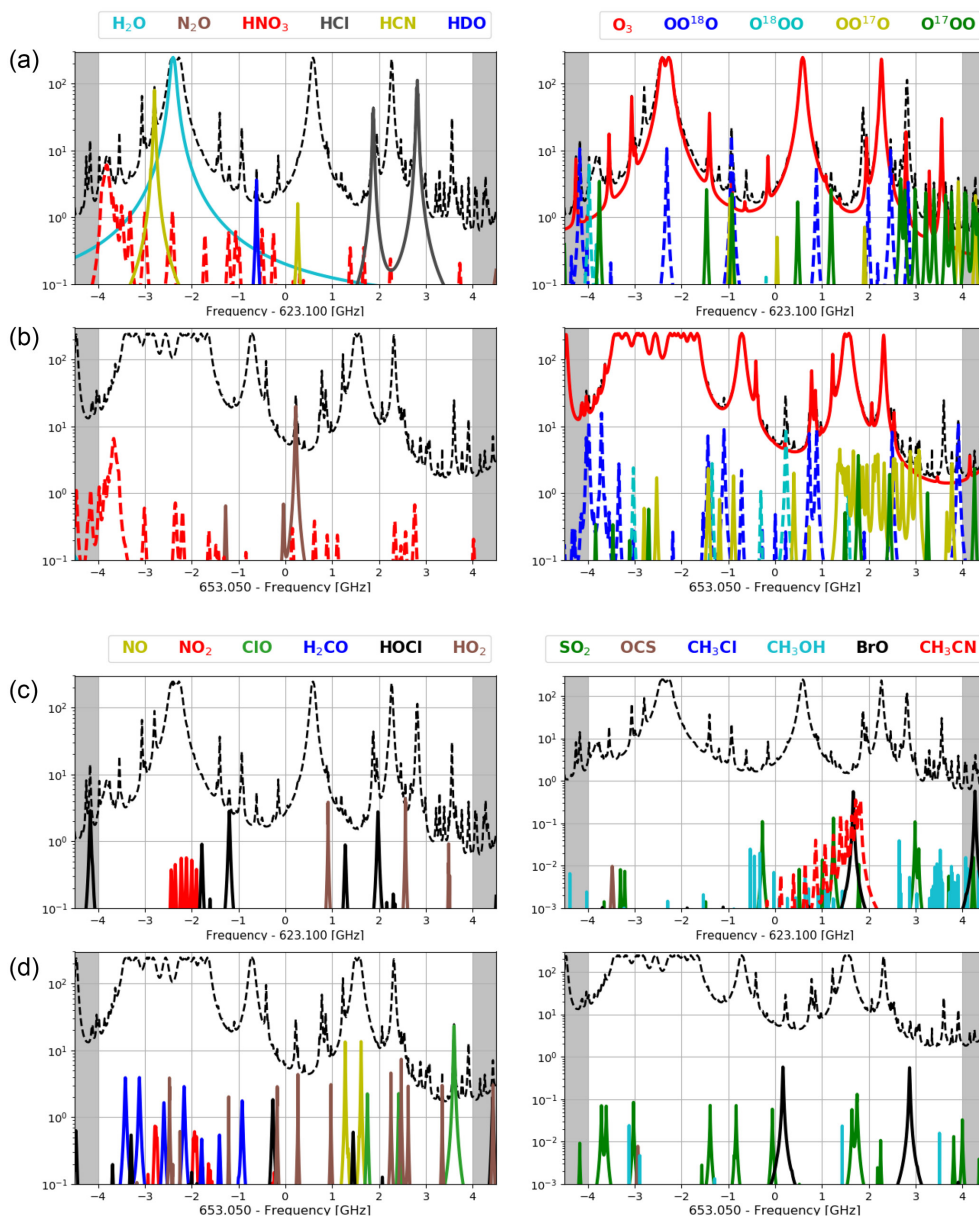


Figure 3. Contributions of the most relevant chemical species to the SIW spectrum. More than 20 molecules are shown in four groups of two panels. In each group, the upper panel shows the lower sideband spectrum (dashed black lines) with a central frequency of 623.1 GHz and the lower panel shows the upper sideband spectrum with a central frequency of 653.05 GHz. The colored lines are single-molecule spectra. Grey-shaded areas are outside the bandwidth. The tangent height is 35 km and frequencies are ordered according to the intermediate frequencies. The intensity is given in brightness temperature (y axis).

factor, used to convert the intensity into brightness temperature:

$$\kappa_b = \frac{c^2}{2k_b \nu_{lo}^2},$$

where $c = 2.997924 \times 10^8 \text{ m s}^{-1}$ is the speed of light in vacuum and $k_b = 1.380662 \times 10^{-23} \text{ J K}^{-1}$ the Boltzmann constant. The spectrometer channel response is assumed to be Gaussian with a full width at half maximum (FWHM) of

1 MHz. Given that the altitude range scanned during the spectrum integration is small compared to the static-antenna vertical resolution (1.1 km and 5 km, respectively), the effective antenna pattern including the vertical scan is approximated by a Gaussian function with the FWHM:

$$\sigma_e^{\text{ant}} = \sqrt{\left(\frac{1.22}{D} \frac{c}{\nu_{lo}}\right)^2 + (\dot{\theta} \Delta T)^2}, \quad (2)$$

where D (m) is the antenna diameter, $\dot{\theta}$ (rad s^{-1}) is the vertical scan velocity and Δ_T is the spectrum integration time. The antenna sidelobes are also neglected. These approximations have negligible impacts on this study. A constant spectral sideband weight is used, $w_{\text{lsb}}(\vartheta) = 0.5$. The integrals in Eq. (1) are computed over ranges $\Delta\theta$ and $\Delta\vartheta$ set to 3 times the FWHM of their corresponding response functions.

3.2 Specific intensity and wind

The specific intensity is computed using the radiative transfer equation:

$$I(\theta, \nu) = \int_{s=0}^{s_r} B_\nu(s) K_\nu(s, \{\nu_a(s)\}_{\text{lines}}) \exp\left(-\int_{s'=s}^{s_r} K_\nu(s', \{\nu_a(s')\}_{\text{lines}}) ds'\right) ds, \quad (3)$$

where s indicates the position on the LOS, B is the Planck function ($\text{W m}^{-2} \text{sr}^{-1} \text{Hz}^{-1}$) and K (m^{-1}) is the absorption coefficient. The background cosmic radiation ($T_b \approx 1 \text{ mK}$) is neglected. The absorption coefficient is computed with a line-by-line model and continua models (Urban et al., 2004). The spectroscopic parameters describing the molecular lines are taken from the HITRAN catalog (Rothman et al., 2009) except those for BrO, CH_3Cl and CH_3CN , which are not available in HITRAN, and are thus taken from the Jet Propulsion Laboratory catalog (Pickett et al., 1998). The frequency of the spectral lines viewed from the receiver ($\{\nu_a(s)\}_{\text{lines}}$) depends on the mean relative motion of the molecules with respect to the receiver, i.e., satellite velocity and wind. The effect of the Doppler on the Planck function is neglected. The line frequency is also shifted by the atmospheric pressure but this effect is small above 25 km where winds are measured. For the H_2O line at 620 GHz, 2 % error in the shift parameter corresponds to an error of 2 m s^{-1} at 10 hPa. The shift on O_3 lines should be smaller but the information is not available in HITRAN and further studies are needed to infer it.

A spherical Earth is assumed for assessing the impacts of all the parameters contributing to the line Doppler shift. At a height z_i and for a LOS nadir angle θ , the line-apparent frequency is (Kursinski et al., 1997)

$$\nu_a(\theta, z_i) = \nu_0 \left(1 - \frac{[V(z_i) \cos(\phi_n) + (U(z_i) + \omega_e R_e \cos(\Lambda)) \sin(\phi_n)] \sin(\alpha_i)}{c} + \frac{W \cos(\alpha_i)}{c} + \frac{V_{\text{sat}} \cos(\phi) \sin \theta}{c} \right), \quad (4)$$

where ν_0 (Hz) is the rest frequency of the transition, V_{sat} is the satellite velocity with respect to a fixed frame attached to

the Earth center, (U, V, W) is the 3-D wind velocity defined with respect to the Earth surface and ω_e , Λ and R_e are the Earth rotation angular velocity (rad s^{-1}), the latitude and the geoid radius at the position i . The LOS nadir angle at z_i is α_i , and ϕ_n is the angle between the LOS and the north direction (Fig. 4).

At the tangent height point ($i = 0$), $\alpha_0 = 90^\circ$ and the Doppler shift $\delta\nu(\theta, z_0)$ is

$$\delta\nu(\theta, z_0) = -\frac{\nu_0}{c} (V_{\text{los}}(z_0) + \omega_e R_e \cos(\Lambda) \sin(\phi_n) - V_{\text{sat}} \cos(\phi) n_0 \frac{z_0 + R_e}{z_r + R_e}), \quad (5)$$

where z_r is the receiver height, n_0 is the refractive index at the tangent point, $\sin(\theta) = n_0 \frac{z_0 + R_e}{z_r + R_e}$ and V_{los} is the LOS component of the horizontal wind:

$$V_{\text{los}}(z_0) = V(z_0) \cos(\phi_n) + U(z_0) \sin(\phi_n). \quad (6)$$

At the Equator and for the forward LOS, the Doppler shifts due to the satellite velocity and to the Earth rotation are $\approx -8 \text{ MHz}$ ($+4000 \text{ m s}^{-1}$) and $\mp 0.74 \text{ MHz}$ ($\pm 370 \text{ m s}^{-1}$), respectively. In order to simplify the calculations, we consider the case of a pseudo-LOS wind profile which, unlike a real one, induces a Doppler shift $\delta\nu(z) = -\nu_0/c V_{\text{plos}}(z)$ that is independent of the angles α_i and Earth rotation, and includes the vertical changes due to the satellite velocity:

$$V_{\text{plos}}(z) = V_{\text{los}}(z) - V_{\text{sat}} \cos(\phi) n_z \left(\frac{z - 50 \text{ km}}{z_r + R_e} \right). \quad (7)$$

At the tangent point, the pseudo-wind induced the same Doppler shift as that given in Eq. (5) to within the same constant over the full vertical scan. The constant includes the Earth rotation effects and most of the satellite velocity ones. The terms embedded in this constant are known with a precision better than 1 m s^{-1} using the star trackers and GPS data onboard the satellite. Such a setting is chosen to yield the satellite-velocity-induced Doppler shift to zero at $z = 50 \text{ km}$, center of the vertical scan.

The pseudo-wind approximation induces errors in the line-apparent frequency at positions on the LOS other than the tangent point. These errors are smaller than 10 cm s^{-1} and have negligible impacts on the retrievals. Indeed, VMR and temperature retrievals are not sensitive to small frequency errors, and regarding wind retrieval, the information is extracted from optically thin measurements which are characterized by narrow specific-intensity weighting functions peaking at the tangent point.

3.3 Calibration and measurement noise

The raw intensity delivered by the spectrometer is expressed as (Olberg et al., 2003)

$$C_{i,j} = G_{i,j} [T_{\text{sys}}(i, j) + \eta_x T_b(i, j) + (1 - \eta_x) T_{\text{so}}(i)], \quad (8)$$

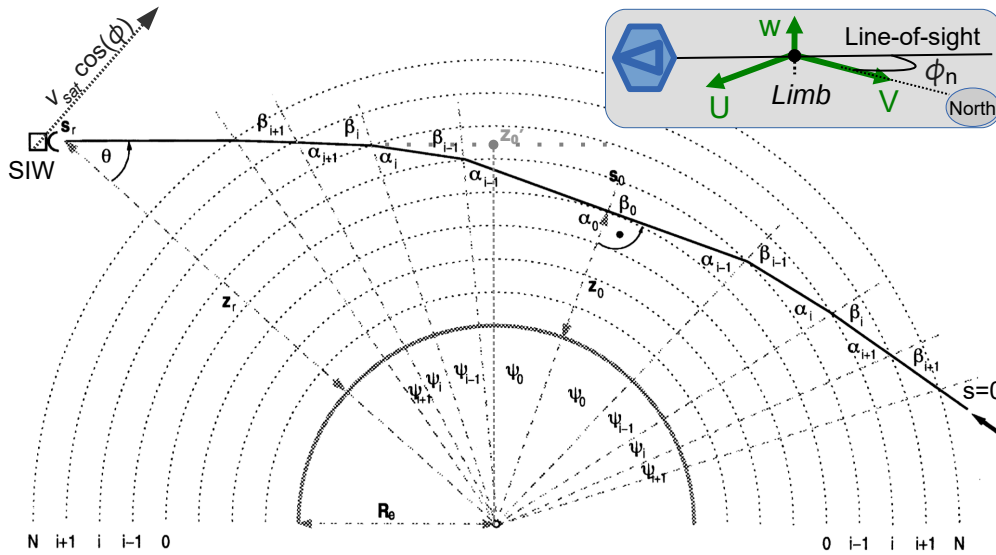


Figure 4. Limb sounding geometry for a refracted line of sight (full line) and a non-refracted one (dashed line). The panel on the right-upper corner shows the orientation of the LOS with respect to the wind components at the tangent point. Figure is adapted from Urban et al. (2004).

where i and j are the tangent height and frequency indices, T_{sys} is the double sideband system temperature, T_{so} is the mean brightness-temperature introduced by the optics spillover, η_x is the efficiency of the integrated antenna ($x = a$) or hot-load horn ($x = c$) and G is the radiometric gain. The last term is

$$G_{i,j} = g_{i,j} (1 - \alpha \langle C_{i,j} \rangle), \quad (9)$$

where $\langle \rangle$ denotes the average over the frequencies j and α is a positive coefficient to account for a non-linear response of the radiometer (Ochiai et al., 2013). The double-sideband system temperature of SIW is expected to be about 1100 K (Omnisys, private communication). The signal intensity is calibrated using the emissions from the cold sky with a Rayleigh–Jeans temperature $T_c \approx 10^{-3}$ K, and from an ambient temperature hot load (Rayleigh–Jeans temperature $T_h \approx 250$ K) measured between two scans. Assuming a linear response of the radiometer and using $T_c \ll T_h$, the radiometer gain is derived as (Olberg et al., 2003)

$$\hat{G}_j = \frac{\bar{C}_h(j) - \bar{C}_c(j)}{\epsilon T_h}, \quad (10)$$

where ϵ is the hot-load emissivity, and C_h and C_c are the receiver raw outputs for the hot load and cold sky. The upper bars $\bar{\cdot}$ indicate that an average value over the whole scan is used. The brightness temperature of the atmospheric signal is then

$$\hat{T}_b(i, j) = \frac{C_{\text{atm}}(i, j) - \bar{C}_c'(i, j)}{\eta_a \hat{G}_{i,j}} + \text{offset}_i, \quad (11)$$

where C_{atm} is the receiver output when the atmosphere is viewed and $\bar{C}_c'(i, j)$ is the cold-sky output interpolated at the

C_{atm} time. We consider that during the scan, the atmosphere and cold sky are viewed alternatively during 0.5 s each. The second term of the equation is a tangent-height-dependent offset induced by the antenna spillover. Such radiance offset is retrieved together with the geophysical information and it is not considered as a retrieval error source. The brightness temperature error from the radiometer noise and the calibration model is

$$\begin{aligned} \delta \hat{T}_b(i, j) &= \frac{\delta C_{\text{atm}}(i, j)}{\hat{G}_j} + \frac{\delta \bar{C}_c'(i, j)}{\hat{G}_j} + \hat{T}_b(i, j) \frac{\delta \hat{G}_j}{\hat{G}_j} \\ &+ \hat{T}_b(i, j) \frac{\delta \eta_a}{\eta_a} + e_{\text{NL}}(i, j) \\ &= \epsilon_{\text{atm}}(i, j) + \epsilon_c'(i, j) + (\epsilon_{\bar{h}}(j) + \epsilon_{\bar{c}}(j)) \\ &\beta_h(i, j) + (e_{\bar{h}} + e_{\eta_a}) \hat{T}_b(i, j) + e_{\text{NL}}(i, j), \end{aligned} \quad (12)$$

where ϵ_{atm} and ϵ_c' are white noises of the atmospheric and cold-sky brightness temperatures (Eq. 11); $\epsilon_{\bar{c}}$ and $\epsilon_{\bar{h}}$ are those of the hot-load and cold-sky spectra in Eq. (10), and $\beta_h = \frac{T_b(i, j)}{\epsilon T_h}$. The two last elements of the equation are systematic errors induced by relative errors $e_{\bar{h}}$ and e_{η_a} on the hot-load emission (ϵT_h) and the antenna efficiency (η_a), and the error due to the receiver non-linearity (e_{NL}).

The noise standard deviation is given by the radiometric equation (Jarnot et al., 2006; Ochiai et al., 2013):

$$\sigma_t(i, j) = \left[T_{\text{sys}}^{\text{dsb}}(i, j) + T_b(i, j) \right] \sqrt{\frac{1}{wt} + \left(\frac{\Delta G}{G} \right)^2}, \quad (13)$$

where w (≈ 1 MHz) is the noise equivalent bandwidth of spectrometer channel and t is the observation time. The

term $1/wt$ describes a spectrally uncorrelated noise, while $(\Delta G/G)$ describes a fully spectrally correlated noise (Jarnot et al., 2006; Ochiai et al., 2013), that is, at first approximation, a radiance offset that is mitigated by the subtraction of the cold sky in Eq. (11). Considering the average and interpolation on the cold-sky and hot-load outputs, the covariance matrix describing the measurement noise is then:

$$S_y(u, u') = \begin{cases} \sigma_a^2 + \sigma_c'^2/2 + (\sigma_c^2 + \sigma_h^2) \beta_h^2(u) & \text{if } i = i' \text{ and } j = j' \\ \sigma_c'^2/2 + (\sigma_c^2 + \sigma_h^2) \beta_h(u) \beta_h(u') & \text{if } |i - i'| = 1 \text{ and } j = j' \\ (\sigma_c^2 + \sigma_h^2) \beta_h(u) \beta_h(u') & \text{if } |i - i'| > 1 \text{ and } j = j' \\ 0 & \text{if } j \neq j' \end{cases}, \quad (14)$$

where $u = i \cdot N_f + j$, $u' = i' \cdot N_f + j'$ and N_f is the number of frequencies per spectrum. Here we consider an integration time of 2 s for the hot-load and cold-sky spectra in Eq. (10) (used for assessing σ_c and σ_h). The time needed for acquiring the hot-load spectra is available between the termination of a scan and the beginning of the next one. Cold-sky spectra can be obtained in very various ways. A simple one is to construct them using the first four and last four cold-sky spectra measured during a scan.

The error e_{NL} due to the radiometer non-linear response, i.e., non-zero α in Eq. (9), is the difference between the true brightness temperature T_b and the calibrated one \hat{T}_b , computed as follows (Baron et al., 2011):

1. $G_{\text{cold},i,j}$ is computed applying Eq. (8) to the cold-sky view assuming $C_c = 1800$ ADU that is consistent with Odin/SMR (Olberg et al., 2003), $T_{\text{sys}} = 1100$ K, $T_b(\text{cold} - \text{sky}) = 0$ K and $\eta_x = 1$. The value $g_{i,j}$ is then computed (Eq. 9).
2. C_{hot} and C_{atm} are computed given $T_{\text{hot}} = 250$ K and T_b using an iterative process initialized with G_{cold} (Eqs. 8 and 9).
3. Finally, we compute \hat{G} (Eq. 10), \hat{T}_b (Eq. 11) and $e_{NL} = \hat{T}_b - T_b$.

4 Retrieval errors

4.1 Reference atmosphere

The measurement performance depends on the atmospheric state, which depends on the latitude, season and local time (for our calculations, we assume that the zonal variations of the mean atmospheric state are negligible). The most relevant parameters to take into account are the profiles of O_3 , H_2O , HCl , temperature and pressure (or geopotential height). A zonal-mean climatology of these parameters has been built,

covering all latitudes divided into 11 bins (Fig. 5). These climatologies are based on Aura MLS observations (v3.3) performed between 15 November 2009 and 15 February 2010. This period has been chosen because of the strong contrast between the winter-pole and summer-pole conditions that provides large meridional variations of atmospheric states. Moreover, it was characterized by a stable northern polar vortex, which was not affected by any strong perturbation (Kutippurath and Nikulin, 2012). MLS observes in the moving direction from a sun-synchronous platform. The orbit inclination is 98° from the equatorial plane. Each latitude is observed at two different local times, e.g., 01:45 and 13:45 LT at the Equator. These two LT are used to characterize the day- and nighttime conditions though it is daytime (nighttime) for both LT over the southern (northern) boreal latitudes (Fig. 2).

Bad data have been removed following the MLS user's guide documentation (Livesey et al., 2011), except for the data flagged with negative errors that are biased toward the MLS retrieval a-priori. Using such data allows us to span the altitude coverage of the profiles up to 110 km with information suitable for this study. Other molecules are taken from the Whole Atmosphere Community Climate Model (WACCM) (Marsh et al., 2013) and extracted at the climatology latitudes and LT. For $HOCl$, HCN and CH_3CN only tropical profiles are used. Because of their relatively weak signal, their variabilities do not impact the overall measurement performance and only typical abundances are needed to discuss the relevance of the measurement.

4.2 Retrieval method

The simulations are performed with the radiative transfer and retrieval codes used in the SMILES research processing chain (Baron et al., 2011) which has been validated with real observations (Kasai et al., 2013). The retrieved state \hat{x} is a vector including all the unknown parameters of the forward model, namely the atmospheric vertical profiles, a radiance offset on each spectrum and a mean pointing angle offset of the whole scan. The atmospheric profiles are the volume mixing ratio (VMR) profiles of the chemical species, as well as those of temperature and LOS wind.

The retrieval altitudes range from 10 to 90 km, a range fully encompassed within the scan range (10–90 km). The grid resolution is 5 km corresponding to the effective vertical field-of-view of the instrument. Such a setting allows us to perform retrievals using a simple linear least-squares method with weak regularization. The retrieved vector is given by the equation:

$$\hat{x} = x_0 + (K^T S_{d,y}^{-1} K + U_x^{-1})^{-1} K^T S_{d,y}^{-1} (y - y_0), \quad (15)$$

where y is the measurement, x_0 is a first guess of the unknown parameters and y_0 is the associated simulated spectra, $K = \frac{\partial T_b}{\partial x}$ is the Jacobian matrix of the forward model (Eq. 1), $S_{d,y}$ is a diagonal matrix equal to the diagonal of S_y (Eq. 14) and U_x is a diagonal matrix for stabilizing the

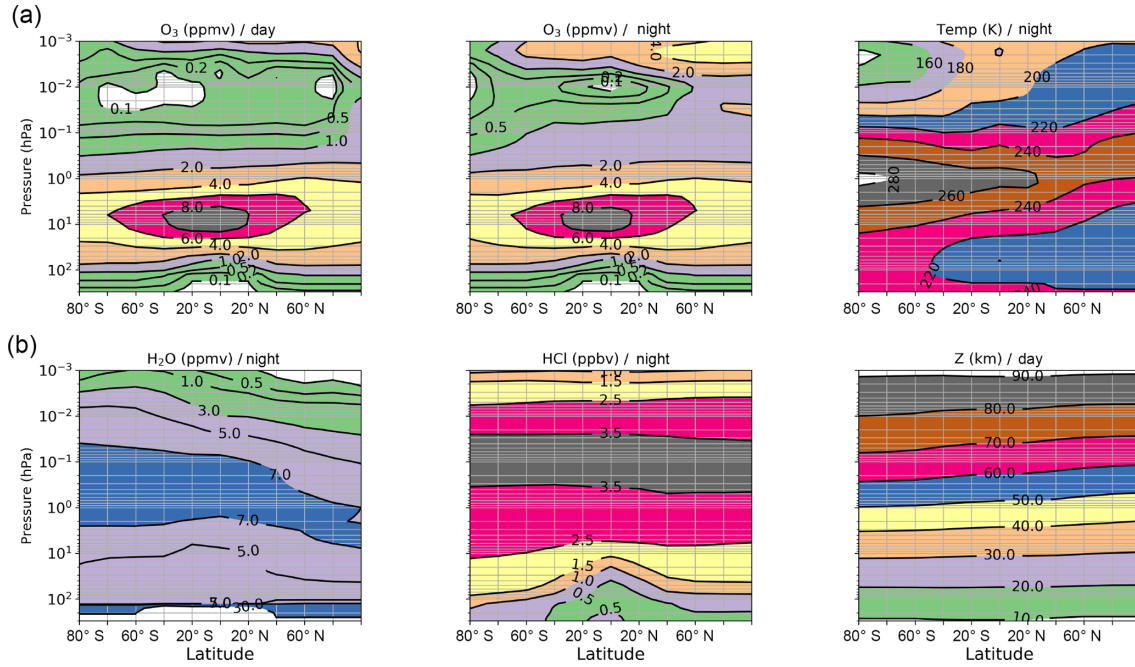


Figure 5. Zonal mean distribution of the most relevant atmospheric parameters for the retrieval error assessment. Images in (a) show the O_3 distributions on pressure levels for day- and night-labeled climatologies as well as the night temperature one. Images in (b) show the night distributions for H_2O , HCl and geopotential altitude. The white regions indicate values smaller than the color scales.

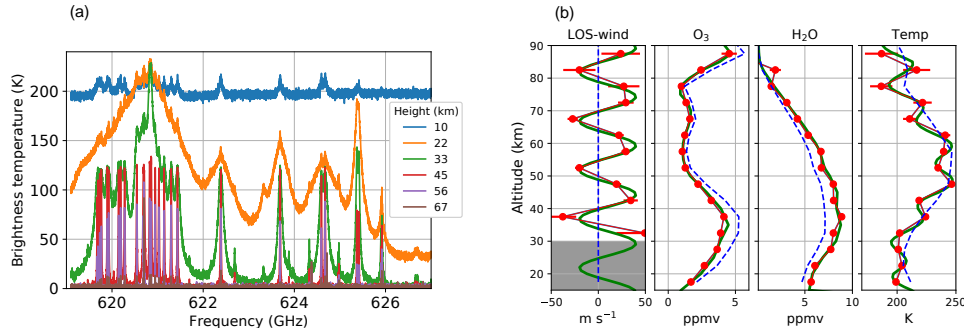


Figure 6. (a) Double sideband spectra with respect to the lower sideband frequency. Only a few spectra of the full vertical scan are shown (see legend). (b) Retrieved profiles with a vertical resolution of 5 km for nighttime arctic conditions. The blue-dashed lines are the a priori profiles (first guess), the green lines are the truth and the red line circles are the retrieved values. The horizontal bars indicate the 1σ errors due to instrument thermal noise.

matrix inversion. Its element square roots correspond to very large standard deviations of \mathbf{x} , typically $> 10000\%$, 1000 K and 1000 m s^{-1} for VMR, temperature and LOS wind, respectively. The regularization effects are negligible where the retrieval errors (null space and measurement errors) are much smaller than the U_x -related STD. In other words, the averaging kernels are unity at altitudes where the measurement is relevant and the retrieval vertical resolution is that of the retrieval altitude grid.

The retrieval precision is derived from the linear mapping of the measurement noise covariance onto the retrieved pa-

rameters space:

$$\epsilon_{x,n}^2 = \text{diag}(\mathbf{G} \mathbf{S}_y \mathbf{G}^T), \quad (16)$$

where $\epsilon_{x,n}$ is the standard deviation of $\hat{\mathbf{x}}$, \mathbf{S}_y is the full measurement covariance matrix (Eq. 14) and $\mathbf{G} = (\mathbf{K}^T \mathbf{S}_{d,y}^{-1} \mathbf{K} + \mathbf{U}_x^{-1})^{-1} \mathbf{K}^T \mathbf{S}_{d,y}^{-1}$.

Figure 6 (right panel) shows retrieved profiles of LOS wind, O_3 , H_2O and temperature using a simulated noisy measurement (Fig. 6, left panel). The measurement is computed using perturbed profiles from the nighttime climatology at 80° N , hereafter named true profiles. The true profiles are defined with a vertical resolution of 0.5 km . The H_2O and

HCl climatological profiles are multiplied by 1.2 and the O₃ one is multiplied by 0.8. An offset of -5 K (10 m s^{-1}) and a 9 km period oscillation with an amplitude of 8 K (15 km , 30 m s^{-1}) are added to the temperature (wind) profile. Good agreement is found between the retrieved and true profiles. Below 40 km, the wind retrieval error strongly increases, and we should consider that 30 km is the lower altitude for wind retrieval. The error increase is due to the pressure broadening of the lines that is about 30–40 MHz at 10 hPa. Other profiles are retrieved with low errors over most of the vertical range. A small oscillation is however seen in the H₂O profile that likely arises from the simple retrieval calculation (linearity and weak regularization). These results are obtained with relatively large differences between the true and reference profiles and show that this retrieval setting can safely be used for the error analysis.

Systematic retrieval errors emerge from uncertainties in the instrument, calibration and forward model parameters and LOS angles (Table 2). It is difficult at this stage of the mission definition to provide proper values for these uncertainties. The given values are relatively close to those expected but rounded in a way that allows for straight-forward linear scaling of retrieval errors according to any future better knowledge of the parameter uncertainties. One may notice that the uncertainty in the line broadening parameter (G_i) is likely underestimated and the actual values should be between 1 and 4 % depending on the line. On the other hand, the calibration parameters are likely overestimated. Anyway these errors induce relatively constant retrieval biases that could be mitigated with ad hoc corrections if their properties are well understood, e.g., timescale and latitudinal variabilities (see for example the JEM/SMILES data analysis in Baron et al., 2013b).

The 24 h variability of the local oscillator frequency is between 2 and 10 kHz which directly results in a LOS wind retrieval uncertainty of $1\text{--}5\text{ m s}^{-1}$. The lower limit corresponds to the scientific requirement and the upper one is the worse acceptable case. Though it is a systematic error, it changes from one scan to another one with a time correlation that has to be determined before launch. The impacts on other retrieved parameters are negligible. Given the proposed design of the instrument (Murtagh, 2016), the 1-year frequency variability may be relatively large ($> 0.5\text{ MHz}$ or 250 m s^{-1}), and we should consider that an absolute frequency knowledge, good enough for retrieving winds, may not be available. Frequency calibration will be performed using short-term wind retrieval bias estimates within 40–60 km where other systematic errors are small.

Retrieval errors from other parameters are investigated in Sect. 5.2 using a perturbation method:

$$\epsilon_{x,p} = G(y_p - y_0), \quad (17)$$

where $\epsilon_{x,p}$ is the error induced by the parameter p and y_p is the measurement assessed after changing the value of p

according to its uncertainty. The spectroscopic errors are expressed for each molecule considering that the errors in the line parameters are mutually independent:

$$\epsilon_{x,M} = \sqrt{\sum_i (\epsilon_{x,M,F_i}^2 + \epsilon_{x,M,G_i}^2 + \epsilon_{x,M,S_i}^2)}, \quad (18)$$

where $\epsilon_{x,M}$ denotes the total spectroscopic error due the molecule M , and F_i , G_i and S_i denote the center frequency, air-broadening parameter and line strength of the line i .

The following errors will not be discussed. The errors in the LOS azimuth and elevation angles induces error smaller than 1 m s^{-1} on the LOS wind retrievals. The mean elevation offset of the scan is retrieved with a precision better than 0.2 mrad. The retrieval error induced by the antenna efficiency is not discussed given that it has similar properties to that induced by the hot-load emission error (Eq. 12).

5 Measurement performance

5.1 Retrieval precision

Results are discussed on pressure levels and the corresponding altitudes are shown in Fig. 5. The precision (1σ) is given for a single scan and a vertical resolution of 5 km. It is possible to use the altitude information inscribed in the pressure broadened lineshape for retrieving atmospheric profile at better resolution but at the cost of the precision. Precision degradation can be afforded for products retrieved from strong signals (e.g., O₃ or temperature) or for those whose the vertical resolution is more scientifically relevant than the temporal or horizontal one (precision can be improved by averaging data). On the other hand, degrading the vertical resolution may be necessary for providing useful information on products derived from weak signals (e.g., HOCl). Later, using the results of this study and based on scientific requirements, the retrieval algorithm will be optimized for providing the best compromise between precision and resolution for each of the main products. Let us note that for all products except for the LOS wind, there are two quasi-simultaneous and quasi-coincident retrievals available from the two LOS (Fig. 2). They can be averaged for improving the precision by a factor of $\sqrt{2}$.

5.1.1 O₃ retrieval

Figure 7 shows the retrieval precisions for temperature, LOS wind, O₃ and H₂O that have the strongest lines. Good precision is found for O₃ retrieval over the whole altitude range (200–0.001 hPa) because of the unusually large number of lines compared to other MM/SMM instruments. Between 100 and 0.2 hPa, the relative error is better than 2 % and does not vary significantly with latitude and local time. High precision of $< 0.4\%$ is found between 50 and 2 hPa. There, the retrieval vertical resolution could be improved to 3–4 km

Table 2. Systematic errors in observational and forward model parameters: calibration hot-load temperature (T_h , Eq. 12) and radiance linearity assumption (α , Eq. 11), sideband ratio (w_{lsb} , Eq. 1), local oscillator frequency (ν_{lo} , Eq. 1) over 24 h and 1 year, antenna efficiency (η_a , Eq. 8), spectroscopic line center frequency (F), pressure broadening (G) and strength (S) and LOS azimuth and elevation angles (θ, ϕ , Fig. 4).

Calibration		Sideband ratio	Local oscillator		Ant. efficiency	Spectroscopy			LOS angles
ϵT_h	α	w_{lsb}	24 h	1 year	η_a	F	G	S	θ, ϕ
1 %	0.5×10^{-5}	0.1 %	2 – 10 kHz	> 0.5 MHz	$\sim \epsilon T_H$	10 kHz	1 %	1 %	0.5 mrad

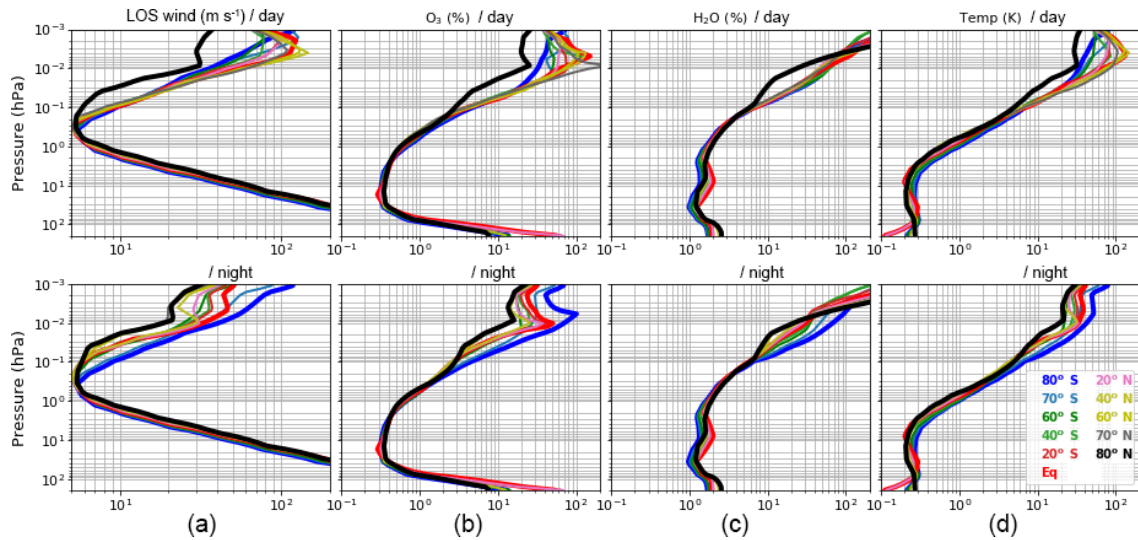


Figure 7. Single-scan retrieval precision (1σ) for line-of-sight wind (a), O_3 (b), H_2O (c) and temperature (d). The line colors correspond to latitude bins (see legend) and thick lines are used for those corresponding to polar and equatorial regions. Errors are given for day- and night-labeled profiles. Note that southern (northern) polar profiles are actually both daytime (nighttime) ones.

with a precision of $\approx 1\%$ (not shown). In the upper part of the retrieval range, the relative precision strongly varies with latitude and local time. The errors are 30–50 % in nighttime and 40–100 % in daytime. The poorest relative precision is found near 0.01 hPa during daytime, where most of O_3 is photo-dissociated (Fig. 5). Above, the relative precision slightly improves due to the O_3 mesospheric secondary peak (Fig. 5).

Figure 8 shows that above 1 hPa most of O_3 information is provided by the first half of the spectrum that contains the cluster of O_3 lines near 655 GHz (Figs. 3 and 6). Below this altitude, both sides of the spectrum contribute equally to the O_3 retrieval. The O_3 line cluster is the main source of information for the LOS wind and temperature retrievals above 4 and 200 hPa, respectively (Fig. 8).

5.1.2 Wind and temperature retrievals

The performance of the LOS wind retrieval strongly depends on the O_3 abundance. With the current definition of the orbit (equatorial ascending node at 18:00 LT), most of the measurements are performed at nighttime (Fig. 2), which is a favorable time for measuring wind. The best performance is

found over the northern polar region, where the nighttime O_3 enhancement is the largest. There, the LOS wind can be retrieved with a precision better than 10 ms^{-1} between 2 and 0.02 hPa (Fig. 7). Comparable performance is found for nighttime equatorial and mid-latitude retrievals over a similar vertical range but with a slightly lower upper limit (0.03–0.04 hPa). In daytime, the uppermost altitude for obtaining similar precision dropped to 0.1 hPa over most of the latitudes. At 10 hPa, the error is $50\text{--}60 \text{ ms}^{-1}$, and averaging 2 weeks of equatorial data in 10° latitude bin gives a precision of about 2 ms^{-1} . Since the precision is much poorer below this altitude, the 10 hPa level should be considered as the lowest altitude for obtaining useful wind information.

At 0.01 hPa, the nighttime LOS wind precision changes with latitude from 20 to 50 ms^{-1} (the southern polar profile is excluded) and from 40 to 60 ms^{-1} in daytime. At this altitude, the H_2O line at 620.7 GHz contributes significantly to the wind retrieval, especially during daytime. Over the polar regions, strong NO enhancements frequently occur in the middle atmosphere due to energetic particle precipitation (EPP) (Randall et al., 2007; Pérot et al., 2014; Orsolini et al., 2017). During such events, the NO lines can be increased by

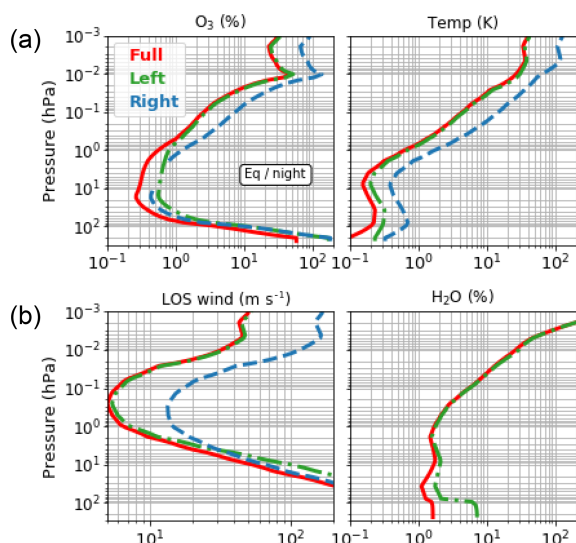


Figure 8. Single-scan retrieval precision (1σ) for O₃ and temperature (a), and line-of-sight wind and H₂O (b). Errors are calculated for a full band retrieval (red lines), the left half IF band (10.075–14.075 GHz, blue lines) and the right half IF band (14.075–18.075 GHz, green lines). Results are shown for equatorial nighttime conditions.

more than a factor of 10, which would improve the wind and temperature retrievals.

Temperature can be retrieved with a precision better than 1 K below 1 hPa. The retrieval vertical resolution can be improved to 3 km with a precision better than 1 K between 200 and 5 hPa (not shown). Above 0.2 hPa, the precision decreases to 10–30 K near 0.01 hPa in nighttime and to 30–80 K in daytime. During daytime most of the mesospheric information is provided by the strong H₂O line at 620.7 GHz.

5.1.3 H₂O and other molecules retrievals

The H₂O profile is retrieved from the line at 620.7 GHz and below 100 hPa, from the continuum induced by far lines. The precision is better than 3 % (20 %) below 0.3 hPa (0.05 hPa). For altitudes above 0.1 hPa, the relative error increases and exhibits large latitudinal variations, e.g., 10–50 % at 0.2 hPa. The largest errors are found during daytime when the signal from O₃ is weak. Under such conditions, temperature is retrieved from the single H₂O line. The forward model inversion becomes ill-conditioned and both H₂O and temperature errors strongly increase. This issue is clearly illustrated with the much smaller H₂O daytime errors estimated without retrieving temperature compared to those with temperature retrieval (Fig. 9). Constraining the mesospheric temperature would significantly improve the mesospheric H₂O retrievals.

The retrieval precision for other molecules are shown in Fig. 9. First, we note that, except for O₃ and H₂O, all chemical species are retrieved from optically thin lines and the VMR error profiles have similar characteristics and are in-

dependent of the VMR values. The minimum VMR error is found near 10 hPa. Below, the errors increase due to spectral line overlapping. The atmosphere becomes opaque near the tropopause. From the middle stratosphere to the mesosphere, the errors increase due to the decrease of atmospheric density (the error is actually proportional to pressure^{-0.6} because the density decrease is partly compensated by the narrowing of the lines).

The best measurement performance with respect to the VMR is found for HCl, N₂O, HCN, CH₃CN and HNO₃. Good information can also be inferred for the four most abundant O₃ isotopologues and from HDO. Important chemically active species such as ClO, NO, NO₂ or HO₂ can also be retrieved. If necessary the relative precision can be improved by averaging profiles or decreasing the retrieval vertical resolution. Deriving useful information for species such as BrO or HOCl will be challenging.

Chemically active species exhibit large variabilities. The photo-chemistry-driven diurnal variation is the most common one. For instance, stratospheric ClO, NO and mesospheric HO₂ are more abundant in daytime but vanish in general during nighttime. Special events that occur more or less frequently can strongly increase the signal-to-noise ratio. For instance, ClO VMR frequently reaches 1.5 ppbv near 20 km during polar springtime due to the chlorine activation during the polar winter. The enhancement of SO₂ after strong volcanic eruption can also be measured (Pumphrey et al., 2015). EPP-induced enhancement of NO_x and HO_x is another example. During such events nighttime NO can reach levels of 10–100 ppbv between 10 and 0.1 hPa, levels much larger than the measurement single-scan precision (2–20 ppbv). EPP-induced enhancements are not well represented in the models (Randall et al., 2007; Pérot et al., 2014; Orsolini et al., 2017), and SIW has strong potential to provide key insights into their dynamical and chemical sources.

5.2 Systematic errors

The errors induced by the spectroscopic uncertainties in the most important lines have been estimated for the LOS wind, temperature, O₃, H₂O and HCl retrievals. We consider the 50 most intense O₃ lines over the whole bandwidth, two HCl triplets (624.9 and 625.9 GHz), two NO triplets (651.4 and 651.7 GHz) and the 620.7 GHz H₂O line. Systematic errors induced by the double-sideband ratio (DSB), the calibration hot-load temperature and the radiometer non-linearity are also discussed for the same products.

5.2.1 Wind retrieval

Figure 10 clearly shows three altitude ranges for the wind retrieval errors induced by the spectroscopic parameters. Results are given for the latitude 60° N. Above 0.1 hPa, a daytime error of 3–5 m s⁻¹ is induced by the frequency uncertainty in the H₂O line (Table A3). During nighttime, the sig-

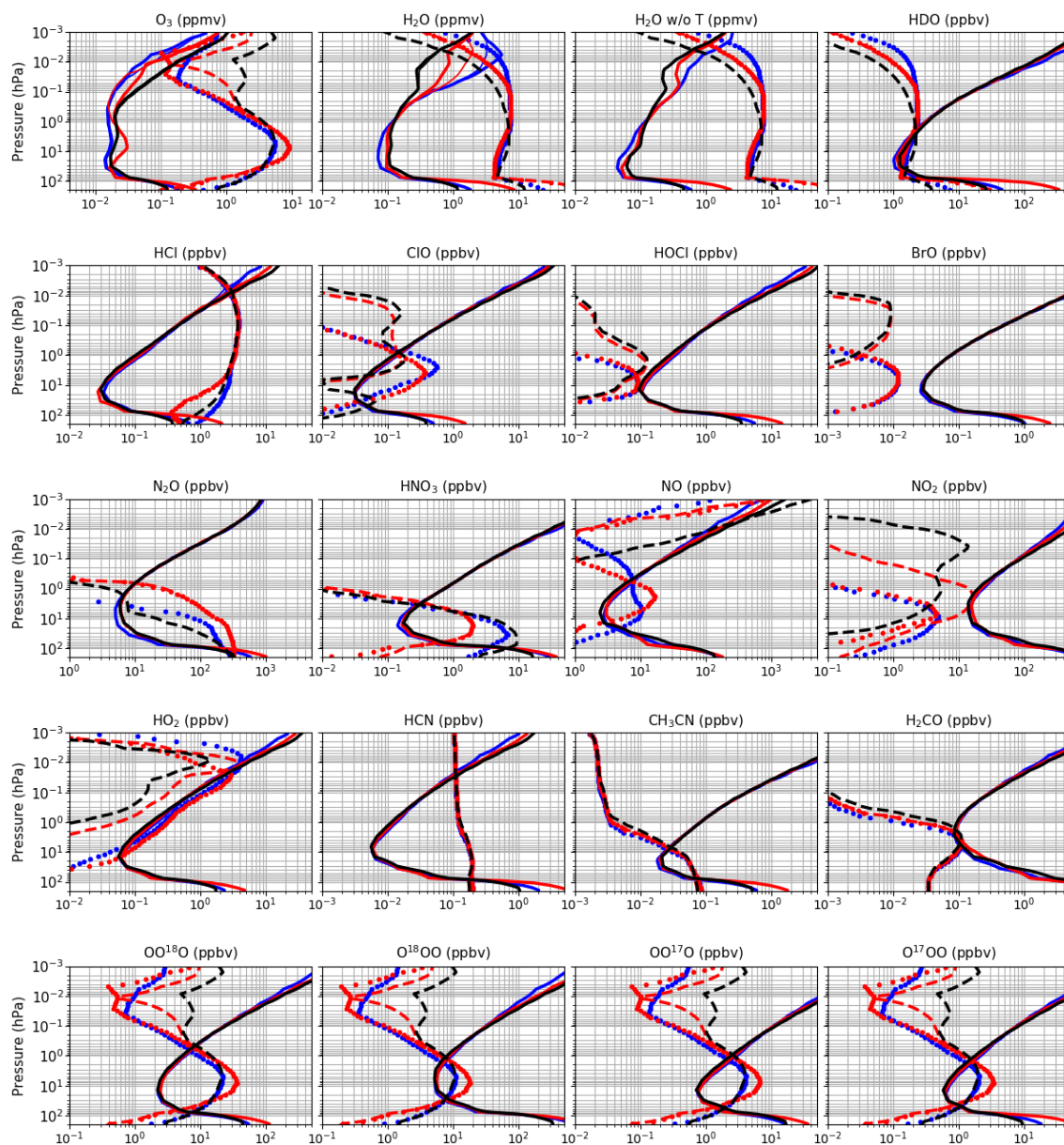


Figure 9. Single-scan retrieval precision (full lines), nighttime VMR (dashed lines) and daytime VMR (dotted lines) profiles. Profiles are shown at 80° S (blue line), Equator (red line) and 80° N (black line). The thick (thin) red full lines are nighttime (daytime) conditions. The H₂O results without temperature retrieval are indicated by “w/o T”.

nal is dominated by about 15 O₃ lines and a retrieval error of only 1 m s⁻¹ is induced by their frequency uncertainty (Tables A1 and A2). The same error of 1 m s⁻¹ is found between 1 and 0.1 hPa both during day and nighttime. No impact of the NO lines has been found, even at higher latitudes, but this is not the case for EPP-enhanced profiles. Below 1 hPa, the lines broadened by the pressure overlap each other. Consequently the uncertainties in the air-broadening parameters and to a lesser extent, the line strength of the O₃ lines contribute to the retrieval error. The bias increases up to 20–30 m s⁻¹ at 10 hPa.

Figure 11 shows the retrieval errors induced by the double-sideband ratio (DSB), the calibration hot-load temperature and the radiometer non-linearity. These parameters introduce errors in the wind retrieval only below 1 hPa. The uncertainty in the calibration hot-load temperature is the dominant retrieval error, reaching 5–8 m s⁻¹ between 2 and 10 hPa. Overall, the O₃ lines parameters are the main source of error between 10 and 1 hPa for all latitudes (Fig. 12).

Methods for mitigating wind retrieval bias have to be developed taking into account SIW observation characteristics (see Appendix A). For JEM/SMILES analysis, a retrieval

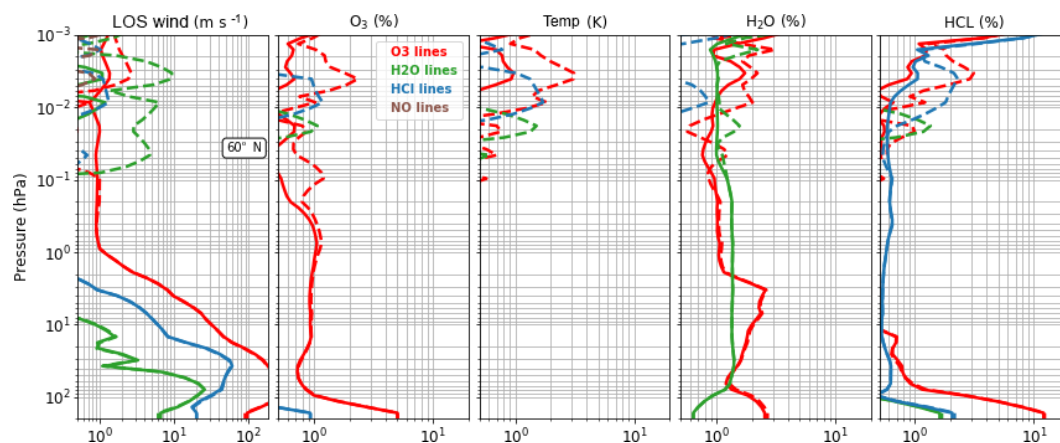


Figure 10. Spectroscopic-induced errors in LOS wind, temperature, O_3 , H_2O and HCl retrievals (see panel titles). The full lines (dashed lines) show the nighttime (daytime) conditions at 60°N .

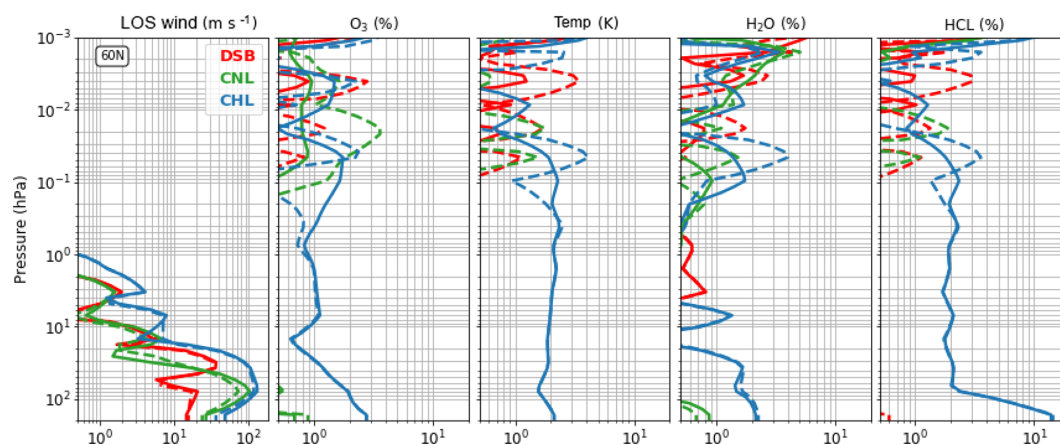


Figure 11. Same as Fig. 10 but for double-sideband ratio (DSB), radiometer nonlinearity (CNL) and calibration hot-load emission (CHL). The error assumptions are summarized in Table 2.

bias of $20\text{--}40\text{ m s}^{-1}$ between 8 and 5 hPa was reduced to less than 4 m s^{-1} between 30°S and 50°N by considering that the mean tropical flow is zonal (Baron et al., 2013b). Meteorological analysis and reanalysis at mid-latitudes can also be used for characterizing the retrieval biases below 5 hPa.

5.2.2 Temperature and VMR retrievals

The biases of O_3 , HCl and temperature retrievals due to the spectroscopic parameters are small. They are lower than 1 % and 0.5 K between 100 and 0.02 hPa. Above 0.02 hPa, the biases increase but remain smaller than 5 % and 4 K. The errors are induced by the air-broadening and strength parameters of O_3 and H_2O lines. The strong impact of the H_2O line parameters onto the HCl retrieval reveals error amplifications due to the temperature retrieval. Using constraints on the temperature retrieval should allow us to reduce such effects.

The retrieval of H_2O above 0.2 hPa has a small bias $< 2\%$ that is induced by the uncertainties in the air-broadening and

strength parameters of the 620.7 GHz H_2O line. Below this altitude, the retrieval error reaches 5 % mainly due to the air-broadening parameters of the overlapping O_3 lines at 620.69 and 623.669 GHz (Table A1). Below 100 hPa, the H_2O lines outside the band are the main signal for the retrieval (not shown), and the 620.7 GHz H_2O line weight in the retrieval is small.

The DSB- and calibration-parameter-induced errors in O_3 , HCl and temperature retrievals are small below 0.1 hPa, i.e., lower than 3 % and 2 K (Fig. 11). The calibration hot-load temperature and the radiometer non-linearity dominate the temperature retrieval error. The VMR and temperature retrievals are also sensitive to the DSB uncertainties and radiometer non-linearity above 0.1 hPa, especially in daytime. The daytime errors are likely increased by the temperature retrieval.

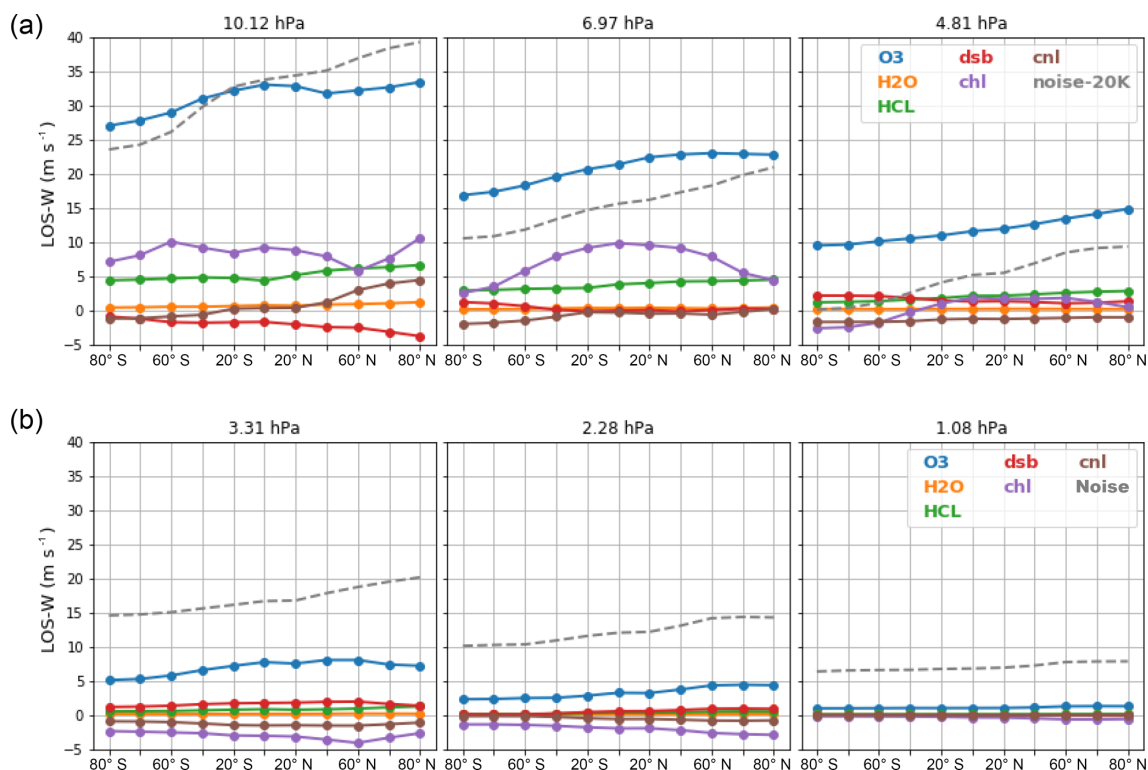


Figure 12. Line-of-sight wind retrieval biases with respect to latitudes near 10, 7 and 5 hPa (a), and near 3, 2 and 1 hPa (b). Biases are shown for the uncertainties in the spectroscopic parameters of O₃, HCl, and H₂O, the double-sideband ratio, the calibration hot load and the calibration non-linearity.

6 Conclusions

A simulation study has been conducted to support the mission definition of SIW and to assess the measurement performance. This small instrument will be launched around 2022 in order to monitor the middle atmosphere (10–90 km) using the thermal emission lines near 640 GHz of a large number of chemical species. This analysis focuses on the main outcomes, namely LOS wind, temperature, O₃ and more than a dozen of other chemical species. The error assessment is performed taking into account the day–night and latitudinal atmospheric variabilities.

The unusually large number of strong O₃ lines at 653–657 GHz allows us to measure the 2-D horizontal wind between 10 and 0.001 hPa and temperature between 100 and 0.1 hPa as well as providing high sensitivity to O₃ between 100 and 0.001 hPa. It is the first time such a mission is designed for measuring horizontal winds. These measurements demand a special observation setting involving two antennas in order to retrieve two perpendicular components of the wind vector. Each component can be measured between 2 and 0.03 hPa with precision better than 10 m s^{−1} and vertical resolution of 5 km. Other spaceborne instruments have poor sensitivity in this altitude range. A sun-synchronous polar orbit allowing us to perform nighttime measurements

is currently being considered. Such conditions are most favorable for mesospheric wind, temperature and ozone measurements but not for active chemical species such as stratospheric ClO or strato-mesospheric HO₂ that generally vanish during nighttime.

The impact of systematic errors induced by the spectroscopic parameters and by the instrument and calibration parameters are discussed. This work highlights the need for good characterization of the spectroscopic parameters (air broadening, strength, center frequency and pressure shift) of key O₃, H₂O and NO lines. Even so, a large wind measurement bias may occur between 10 and 2 hPa mainly due to errors in O₃ line air-broadening parameters. Hence ad hoc methods for reducing retrieval biases must be studied. These methods can be used to calibrate the long-term LO frequency trend that may arise with the proposed hardware. However, improvements of the instrument design for following the LO frequency with a precision better than 2 kHz are still being investigated.

SIW shows strong potential for the study of various scientific questions. It can provide for the first time global information on the horizontal wind between 30 and 90 km that can be used to validate chemical and climate models. It has the potential to contribute to the characterization of long trend series of temperature, O₃, H₂O and HCl, which are important

for climate studies and for monitoring the chemical composition of the middle atmosphere. The mission can provide data to study the dynamics of the middle atmosphere. Based on SIW observations, it will be possible to carry out specific studies on key dynamical processes, such as the quasi-biennial oscillation, the semi-annual oscillation or sudden stratospheric events for example. A better understanding of these phenomena, in addition to global middle-atmospheric wind measurements, would significantly improve our knowledge of the climate system. Not discussed in this paper is the capability of SIW for measuring ice water content in the tropical upper troposphere (Eriksson et al., 2014). Observing the same air mass from two perpendicular directions could provide interesting information considering the high spatial inhomogeneities of cloudy conditions.

Optimization of the calibration procedure will be studied in order to improve the measurement precision. Here we have assumed an equal observation time for the cold-sky and atmosphere measurements. Changing the time allotment in favor of atmospheric observations could improve measurement sensitivity by more than 20 %. The retrievals could also be improved in the mesosphere by increasing the frequency resolution to 0.5 MHz at the intermediate frequency range between 17.2 and 17.4 GHz that contains the strong H₂O line (620.701 GHz) and the two strongest O₃ lines (620.825 and 655.289 GHz). Implementing such a setting is under investigation.

The InnoSat platform offers a rapidly approaching opportunity to fly SIW. This is important since current SMM limb instruments have already by far exceeded their expected lifetimes and are at risk of stopping operation in the near future. However such a platform strongly limits the design of an SMM instrument and its performance. A larger antenna would improve the vertical resolution and an additional receiver with narrow bandwidth measuring an oxygen or a strong water vapor line would significantly improve the wind and temperature retrievals in the mesosphere. Such improvements are being considered for the much larger SMILES-2 instrument (Ochiai et al., 2017) presented in the introduction section and which also includes the same spectral window as SIW. However this instrument cannot be launched before 2025 if decided upon.

Data availability. Model and data are available upon request .

Appendix A: LOS and horizontal winds

The retrieval of two line-of-sight winds over the same region allows us to compute the meridional (V) and zonal (U) components of the horizontal wind vector. Applying Eq. (6) to the forward and aftward viewing antenna, the two retrieved LOS winds are

$$\begin{aligned} V_{\text{los,fwd}} &= U \sin(\phi_n) + V \cos(\phi_n), \\ V_{\text{los,aft}} &= U \sin(\phi_n + \delta) + V \cos(\phi_n + \delta), \end{aligned} \quad (\text{A1})$$

where ϕ_n is the angle of the forward-looking line of sight with respect to the north direction and δ is the angle between the two lines of sight. It is straight forward to show that

$$\begin{aligned} U &= \frac{1}{\sin(\delta)} (V_{\text{los,aft}} \cos(\phi_n) - V_{\text{los,fwd}} \cos(\phi_n + \delta)), \\ V &= \frac{1}{\sin(\delta)} (V_{\text{los,fwd}} \sin(\phi_n + \delta) - V_{\text{los,aft}} \sin(\phi_n)), \end{aligned} \quad (\text{A2})$$

and the random errors in U and V are

$$\begin{aligned} \epsilon_U &= \frac{\epsilon_{\text{los}}}{\sin(\delta)} \sqrt{\cos(\phi_n + \delta)^2 + \cos(\phi_n)^2}, \\ \epsilon_V &= \frac{\epsilon_{\text{los}}}{\sin(\delta)} \sqrt{\sin(\phi_n + \delta)^2 + \sin(\phi_n)^2}, \end{aligned} \quad (\text{A3})$$

where ϵ_{los} is line-of-sight wind retrieval error.

For $\delta = 90^\circ$, we have

$$\begin{aligned} U &= (V_{\text{los,aft}} \cos(\phi_n) + V_{\text{los,fwd}} \sin(\phi_n)), \\ V &= (V_{\text{los,fwd}} \cos(\phi_n) - V_{\text{los,aft}} \sin(\phi_n)). \end{aligned}$$

The random error in each wind component becomes $\epsilon_U = \epsilon_V = \epsilon_{\text{los}}$. This configuration is that for which $\epsilon_U^2 + \epsilon_V^2 = 2(\epsilon_{\text{los}}/\sin(\delta))^2$ is minimum.

A systematic error e_{los} on the LOS wind retrievals propagates to the U and V components as follows:

1. the systematic error in the zonal wind estimate is $e_u = e_{\text{los}} (\cos(\phi_n) + \sin(\phi_n))$;
2. the systematic error in the meridional wind estimate is $e_v = e_{\text{los}} (\cos(\phi_n) - \sin(\phi_n))$.

We assume that e_{los} does not depends on the LOS orientation which is a valid assumption for the errors investigated in this paper (LO frequency, calibration, spectroscopy). We should note that $e_v = 0$ for $\phi_n = 45^\circ$ or 225° , which occurs at latitudes between 30 and 50° N on the ascending branch of the orbit and between 10 and 30° N on the descending branch. The cases $e_u = 0$ occur for the measurements from the orbit positions with the lowest and highest latitudes.

At the Equator, the bias of the meridional wind is partly canceled out. For instance, an error $e_{\text{los}} = 1.0 \text{ m s}^{-1}$ induces an error $e_v = 0.2 \text{ m s}^{-1}$. Hence, using the tropical zonal wind to characterize the measurement bias as it was done in the JEM/SMILES analysis may not be a satisfactory method for SIW. On the other hand, $e_{\text{los}} = 1.0 \text{ m s}^{-1}$ leads to an error $e_u = 1.4 \text{ m s}^{-1}$ on the zonal component with an opposite sign on the ascending and descending orbit branches. The sign difference may provide us with a way to characterize LOS wind retrieval systematic errors.

Appendix B: Spectroscopic lines

The following tables show the most relevant spectroscopic lines for the retrievals of the LOS wind, O_3 , temperature, H_2O and HCl . The relative retrieval impact of each parameter is defined as

$$Q_{x,M,p_i} = \frac{\epsilon_{x,M,p_i}}{\max(\{\epsilon_{x,M,p_i}\}_{p,i})} \text{ with } p_i = F_i, G_i \text{ or } S_i, \quad (\text{B1})$$

where M denotes the chemical species, ϵ_{x,M,p_i} is the error induced by the parameter p of the line i (Eq. 18) and $\{\epsilon_{x,M,p_i}\}_{p,i}$ is the set of errors induced by all the parameters of all the lines of the species M .

Table A1. Relative impact of O₃ line parameters on temperature, O₃, H₂O and LOS wind retrievals. For a retrieved product, the impact is defined as the ratio of the error to the maximum error (see text). Results are given for 10, 1 and 0.1 hPa levels (equatorial nighttime conditions). Only parameters having an impact larger than 0.5 at any of the considered altitudes are shown. The parameters are the center frequency (*S*), the air-broadening parameter (*G*) and the line strength (*S*). The line is characterized by its frequency (MHz).

Parameter	LOS wind			O ₃			Temperature			H ₂ O			HCl		
lower sideband															
620687- <i>F</i>	–	–	0.6	–	–	–	–	–	–	–	–	–	–	–	–
- <i>G</i>	–	–0.5	–	–	–	–	–	–	–	0.8	0.5	–	–	–	–
- <i>S</i>	–	– 1.0	–	–	–	–	–	–	–0.5	1.0	0.7	–	–	–	–
620825- <i>F</i>	–	0.6	1.0	–	–	–	–	–	–	–	–	–	–	–	–
- <i>G</i>	0.8	–	–	–	–	–	–	0.9	–	0.7	–	–	–	–	–
- <i>S</i>	0.7	–	–	–	–	–	–	1.0	0.8	0.6	–	–	–	–	–
623688- <i>F</i>	–	0.8	0.7	–	–	–	–	–	–	–	–	–	–	–	–
- <i>G</i>	–	–	–	–	–	–	0.8	–	–	–0.7	–0.5	–	–	–	–
- <i>S</i>	–	–	–	–	–	–	1.0	–	–	–0.8	–0.5	–	–	–	–
625370- <i>F</i>	–	0.7	–	–	–	–	–	–	–	–	–	–	–	–	–
- <i>G</i>	–1.0	–	–	–	–	–	–	–	–	–	–	–	–	–	–
- <i>S</i>	–	–	–	1.0	1.0	1.0	– 1.0	–	0.5	–	1.0	1.0	–	1.0	1.0
upper sideband															
650732- <i>F</i>	–	0.7	–	–	–	–	–	–	–	–	–	–	–	–	–
- <i>G</i>	– 1.0	–	–	–	–	–	–	–	–	–	–	–	–	–	–
- <i>S</i>	–0.5	–	–	0.9	0.9	0.9	–0.9	–	–	–	0.9	0.9	–	1.0	0.9
651475- <i>F</i>	–	0.7	1.0	–	–	–	–	–	–	–	–	–	–	–	–
- <i>G</i>	–0.5	–	–	–	–	–	–	–	–	–	–	–	1.0	–	–
- <i>S</i>	–0.6	–	–	–	–	–	–	–	–	–0.5	–	–	1.0	–	–
651556- <i>F</i>	–	0.9	0.5	–	–	–	–	–	–	–	–	–	–	–	–
- <i>S</i>	–	–	–	–	–	–	–	–	–0.9	–	–	–	–	–	–
653763- <i>F</i>	–	0.7	0.9	–	–	–	–	–	–	–	–	–	–	–	–
- <i>G</i>	–	–	–	–	–	–	–	–	–	–	–	–	–0.6	–	–
- <i>S</i>	–	–	–	–	–	–	–	–	–	–	–	–	–0.6	–	–

Table A2. Continuation of Table A1.

Parameter	LOS wind			O ₃			Temperature			H ₂ O			HCl		
654713- <i>F</i>	–	0.7	0.9	–	–	–	–	–	–	–	–	–	–	–	–
- <i>S</i>	–0.5	–	–	–	–	–	–	–	–	–	–	–	–	–	–
654851- <i>F</i>	–	0.7	0.9	–	–	–	–	–	–	–	–	–	–	–	–
655004- <i>F</i>	–	0.7	0.8	–	–	–	–	–	–	–	–	–	–	–	–
655121- <i>F</i>	–	0.7	0.8	–	–	–	–	–	–	–	–	–	–	–	–
655202- <i>F</i>	–	0.7	0.9	–	–	–	–	–	–	–	–	–	–	–	–
655289- <i>F</i>	–	0.6	1.0	–	–	–	–	–	–	–	–	–	–	–	–
- <i>G</i>	–	–	–	–	–	–	–	0.5	–	–	–	–	–	–	–
- <i>S</i>	–	–	–	–	–	–	–	0.6	0.8	–	–	–	–	–	–
655607- <i>F</i>	–	0.7	0.9	–	–	–	–	–	–	–	–	–	–	–	–
- <i>S</i>	–	–	–	–	–	–	–	–	–	0.5	–	–	–	–	–
655873- <i>F</i>	–	0.7	0.9	–	–	–	–	–	–	–	–	–	–	–	–
655960- <i>F</i>	–	0.7	0.9	–	–	–	–	–	–	–	–	–	–	–	–
656005- <i>F</i>	–	0.8	0.8	–	–	–	–	–	–	–	–	–	–	–	–
- <i>S</i>	–	–	–	–	–	–	–	–	0.5	–	–	–	–	–	–
656224- <i>F</i>	–	0.7	0.8	–	–	–	–	–	–	–	–	–	–	–	–
656251- <i>F</i>	–	0.7	0.7	–	–	–	–	–	–	–	–	–	–	–	–
656383- <i>F</i>	–	0.8	0.6	–	–	–	–	–	–	–	–	–	–	–	–
- <i>S</i>	–	–	–	–	–	–	–	–	–0.6	–	–	–	–	–	–
656419- <i>F</i>	–	0.8	0.5	–	–	–	–	–	–	–	–	–	–	–	–
- <i>S</i>	–	–	–	–	–	–	–	–	–0.9	–	–	–	–	–	–
656461- <i>F</i>	–	0.7	–	–	–	–	–	–	–	–	–	–	–	–	–
- <i>S</i>	–	–	–	–	–	–	–	–	–1.0	–	–	–	–	–	–
656476- <i>F</i>	–	0.6	–	–	–	–	–	–	–	–	–	–	–	–	–
- <i>S</i>	–	–	–	–	–	–	–	–	–0.7	–	–	–	–	–	–

Table A3. Same as Table A1 but for the H₂O line parameters

Parameter	LOS wind			O ₃			Temperature			H ₂ O			HCl		
620701- <i>F</i>	–	1.0	1.0	–	–1.0	–	–	–	–	–	–	–	–	0.5	–
- <i>G</i>	1.0	–1.0	–	0.9	0.8	1.0	0.6	–	1.0	0.9	0.9	0.7	0.7	–	1.0
- <i>S</i>	–	–	–	1.0	0.7	–	1.0	1.0	–	1.0	1.0	1.0	1.0	1.0	–
622482- <i>G</i>	–	–	–	–	–0.5	–	–	–	–	–	–	–	–	–	–
- <i>S</i>	–0.8	–	–	–	0.7	–	–	–	–	–	–	–	–	–	–

Table A4. Same as Table A1 but for the HCl line parameters.

Parameter	LOS wind			O ₃			Temperature			H ₂ O			HCl		
624964- <i>S</i>	–0.6	–	–	–	–	–	–0.6	0.5	–	–	–	–	–	–	–
624977- <i>G</i>	–	–	–	–	–	–	0.6	–	–	–	–	–	–	–	–
- <i>S</i>	–	–	–	–	–	–	–1.0	0.5	–	–	–	–	–	–	–
625901- <i>F</i>	–	0.6	0.6	–	–	–	–	–	–	–	–	–	–	–	–
- <i>G</i>	0.9	–	–	–0.7	–0.7	–0.6	–0.6	–0.9	–	–	–0.6	–0.6	–	–	–
- <i>S</i>	–1.0	–	–	–	–	–	1.0	–	0.5	–	–	–	0.7	0.5	0.6
625918- <i>F</i>	–	1.0	1.0	–	–	–	–	–	0.5	–	–	–	–	–	–
- <i>G</i>	–	–	–	–1.0	–1.0	–1.0	–	–1.0	1.0	–1.0	–1.0	–1.0	–	–0.5	–
- <i>S</i>	1.0	–	–	–	–	–	0.6	–1.0	0.6	0.7	–	–	1.0	1.0	1.0
625931- <i>G</i>	–0.6	–	–	–	–	–	–	–0.5	–	–	–	–	–	–	–
- <i>S</i>	1.0	–	–	–	–	–	–	–	–0.7	–	–	–	–	–	–

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Competing interests. The authors declare that they have no conflict of interest

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